Notification of a Proposal to issue an Airworthiness Directive

PAD No.: 26-129

Issued: 21 October 2026

Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below.
All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation date indicated.

Design Approval Holder's Name:
AIRBUS S.A.S.

Type/Model designation(s):
A350 aeroplanes

Effective Date: [TBD - standard: 14 days after AD issue date]

TCDS Number(s): EASA.A.151

Foreign AD: Not applicable

Supersedure: This AD supersedes EASA AD 2025-0008 dated 09 January 2025.

ATA 27 – Flight Controls – Flight Control Remote Module – Replacement / Life Limitation

---

Manufacturer(s):
Airbus

Applicability:
Airbus A350-941 and A350-1041 aeroplanes, all manufacturer serial numbers.

Definitions:
For the purpose of this AD, the following definitions apply:

Affected part: Flight Control Remote Module (FCRM), having Part Number (P/N) CA71323-013, P/N CA71323-014, P/N CA71323-015 or P/N CA71323-016.

Serviceable part: An FCRM, eligible for installation in accordance with Airbus instructions, which is not an affected part, or an affected part that has accumulated less than 5 500 flight cycles (FC) and less than 32 000 flight hours (FH) since new (first installation on an aeroplane) (see Note 1 of this AD).

The SB: Airbus Service Bulletin (SB) A350-27-P066 Revision 1.

An agency of the European Union

TE.CAP.00112-010 © European Union Aviation Safety Agency. All rights reserved. ISO9001 Certified.
Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

Reason:

Occurrences of electronic centralised aircraft monitor (ECAM) messages requiring flight control remote module (FCRM) replacement have been reported. Further investigation identified an issue linked to solder structural fatigue.

This condition, if not corrected, could lead to failure of a flight control actuator, possibly resulting in reduced control of the aeroplane.

To address this potential unsafe condition, Airbus issued the SB at original issue, to introduce a life limit for the affected parts and to provide instructions for determination of their accumulated FH and FC. Consequently, EASA issued AD 2025-0008 to require replacement of affected parts before exceeding the life limit.

Since that AD was issued, it has been determined that the life limit must be reduced. The SB has been published to update the life limit of the affected parts.

For the reason described above, this AD partially retains the requirements of the EASA AD 2025-0008, which is superseded, and introduces reduced life limit for the affected parts.

Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

Part Replacement:

(1) Within the compliance time as provided in Table 1 of this AD, replace each affected part with a serviceable part, as defined in this AD, in accordance with the instructions of the SB.

Note 1: The FC and the FH referred in paragraph (1) of this AD are those accumulated by the affected parts since new (first installation on an aeroplane). If unknown, the SB provides instructions to determine the FC and FH accumulated by an affected part since new.

Table 1 - Affected Parts Replacement

	Compliance Time: A or B, whichever occurs later
A	Before the affected part accumulates 5 500 FC or 32 000 FH (see Note 1 of this AD), whichever occurs first
B	Within 2 years and 5 months after the effective date of this AD

Part(s) Installation:

(2) From the effective date of this AD, it is allowed to install an affected part on any aeroplane, provided the part is a serviceable part, and that, following installation, the affected part is replaced as required by paragraph (1) of this AD (see Note 2 of this AD).

An agency of the European Union

TE.CAP.00112-010 © European Union Aviation Safety Agency. All rights reserved. ISO9001 Certified.
Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

Note 2: Removal of an affected part and reinstallation of that affected part at the same location on the same aeroplane, accomplished during a single maintenance visit, is not considered as installation as specified by paragraph (2) of this AD.

Ref. Publications:

Airbus SB A350-27-P066 original issue dated 12 November 2024, or Revision 01 dated 05 June 2026.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

1. This Proposed AD will be closed for consultation on 18 November 2026.

2. Enquiries regarding this PAD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.

3. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this PAD, and which may occur, or have occurred on a product, part or appliance not affected by this PAD, can be reported to the EU aviation safety reporting system. This may include reporting on the same or similar components, other than those covered by the design to which this PAD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.

4. For any question concerning the technical content of the requirements in this PAD, please contact: AIRBUS A350 XWB (1IAK), E-mail: continued-airworthiness.a350@airbus.com.

An agency of the European Union

TE.CAP.00112-010 © European Union Aviation Safety Agency. All rights reserved. ISO9001 Certified.
Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.