Notification of a Proposal to issue an Airworthiness Directive

PAD No.: 26-138

Issued: 13 November 2026

Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below.
All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation date indicated.

Design Approval Holder's Name:
AIRBUS S.A.S.

Type/Model designation(s):
A330 aeroplanes

Effective Date: [TBD - standard: 14 days after AD issue date]

TCDS Number(s): EASA.A.004

Foreign AD: Not applicable

Supersedure: None

ATA 71 – Powerplant – Engine Mounts – Inspection

Manufacturer(s):
Airbus, formerly Airbus Industrie

Applicability:
Airbus A330-201, A330-202, A330-203, A330-301, A330-302 and A330-303 aeroplanes, all manufacturer serial numbers.

Definitions:
For the purpose of this AD, the following definitions apply:

The SB: Airbus Service Bulletin (SB) A330-71-3045.

Affected part: Engine forward mount platform, having part number 1843M81G01, and a serial number as listed in Appendix 1 of this AD, except those in any of the following conditions:

- Have passed a fluorescent penetrant inspection (FPI) in accordance with General Electric (GE) instructions on 01 January 2025 or later; or
- Have been last inspected (FPI) at a maintenance facility other than GE Evergreen Engine Services (GEEVES).

Affected engine: An engine having an affected part installed.

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Serviceable engine: Any engine, eligible for installation in accordance with Airbus instructions, which is not an affected engine.

Groups: Group 1 aeroplanes are those that have an affected engine installed.
Group 2 aeroplanes are those that do not have an affected engine installed.

Reason:

A quality investigation determined that, between 2020 and 2024, FPI of affected parts may have not been properly accomplished at GEEVES, possibly leading to an increase of the minimum detectable crack length.

This condition, if not detected and corrected, could affect the effectiveness of the applicable Airworthiness Limitation Item task, possibly leading to failure of an affected part and in-flight separation of the engine from the aeroplane.

To address this potential unsafe condition, GE issued SB CF6-80E1 72-0609, providing instructions to accomplish an in-shop FPI of affected parts, and Airbus issued the SB, providing instructions to remove affected engines and to accomplish an FPI of affected parts in accordance with the instruction of the GE SB.

For the reasons described above, this AD requires replacement of affected engines with serviceable engines.

Required Action(s) and Compliance Time(s):

Required as indicated, by this AD, unless the action(s) required by this AD have been already accomplished:

Replacement:

(1) For Group 1 aeroplanes: Before an affected part accumulates 4 700 flight cycles since last accomplishment of an FPI, replace the engine having that affected part installed with a serviceable engine, as defined in this AD (see Note 1 of this AD).
The applicable Aircraft Maintenance Manual provides instructions which are acceptable to comply with this requirement.

Note 1: The SB provides instructions for an in-shop FPI of the affected part of an engine. Following accomplishment of that FPI and, as applicable, corrective actions, that engine is considered to be a serviceable engine.

Part Installation(s):

(2) For Group 1 and Group 2 aeroplanes: From the effective date of this AD, it is allowed to install an affected engine on any aeroplane, provided that, following that installation, that engine is replaced as required by paragraph (1) of this AD.

Ref. Publications:

Airbus SB A330-71-3045 original issue dated 02 October 2026.

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The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

1. This Proposed AD will be closed for consultation on 11 December 2026.

2. Enquiries regarding this PAD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.

3. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this PAD, and which may occur, or have occurred on a product, part or appliance not affected by this PAD, can be reported to the EU aviation safety reporting system. This may include reporting on the same or similar components, other than those covered by the design to which this PAD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.

4. For any question concerning the technical content of the requirements in this PAD, please contact: AIRBUS – 1IAL (Airworthiness Office), E-mail: airworthiness.A330-A340@airbus.com.

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Appendix 1 – Affected parts

Part Number	Serial Number
1843M81G01	CDAAF682, COAAPJ9T, CDAAPW6R, TMT5JM80, MMTE8991, MMTE4272, CDAA7896, CDAAPARA, CDAAK782, CDAAA843, CDAAB901, CDAAB914, CDAAR5YN, CDAAC762, CDAAA857, CDAAE160, CDAAE158, CDAAE155, CDAAE711, CDAAA851, CDAAPTMF, CDAAPARB, CDAAP4FS, CDAAPDT1, CDAAPPJ2, CDAAPPHW, CDAAPW6U, CDAA9696, CDAAR24G, CDAAR24J, TMT5JK20, TMT5JK32, TMT5JK35, TMT5JK58, TMT5JK57, TMT5JK61, TMT5JK64, TMT5JK74, TMT5JK76, TMT5JM32, TMT5JK85, TMT5JK44, TMT5JK87, TMT5JK91, CDAA2599, TMT5JM07, TMT5JM13, TMT5JM20, TMT5JK88, TMT5JM44, TMT5JM54, TMT5JM56, TMT5JM59, FRD00226, TMT5JM30, FRD00230, TMT5JM51, TMT5JM75, TMT5JM78, TMT5JM69, FRD00312, TMT5JM57, CDAAA853, CDAA5362, CDAAP4G3, CDAAE709, CDAAF687, CDAAP0C5, CDAAPARH, CDAAF683, CDAAPPJ0, CDAAPPJ3, CDAAPTMC, CDAAR126, CDAAPTM7, CDAAR24H, TMT5JK05, TMT5JK01, TMT5JK38, TMT5JK29, TMT5JK34, TMT5JK55, TMT5JK54, TMT5JK60, TMT5JK72, TMT5JK59, TMT5JK69, TMT5JK73, TMT5JK70, TMT5JK68, TMT5JK84, TMT5JK82, TMT5JK89, TMT5JK99, TMT5JM10, TMT5JM22, TMT5JM23, TMT5JM26, TMT5JM25, TMT5JM41, TMT5JM45, TMT5JM43, TMT5JM46, TMT5JM49 TMT5JM52, TMT5JM58, TMT5JM60, FRD00229, TMT5JM65, CDAAC757, CDAAK779, TMT5JK43, TMT5JM19, FRD00227, FRD00228

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